

ELECTRICAL POWER SYSTEM (EPS)

MAIN FFATURES

- Flight heritage acquired in 2021 with the launch of RadCube
- Cold redundancy and graceful degradation
- Single point failure tolerant
- Telemetry and telecommand communication with redundant CAN bus
- Temperature ranges:
- circuits: -40°C to 80°C qualified
- battery: 0°C to 50°C
- Autonomous LCL protection
- Autonomous battery heater and balancing
- Two-level battery undervoltage protection
- Robust aluminium (T-7070) milling case
- Recommended for 3U/6U and 12U platforms
- Radiation tested (TID)



SPECIFICATION

Sa	lar	Energy	Harv	actina

Solar Energy Harvesting	
Solar Cell Type	Triple Junction Solar Cells, (~30% eff.)
Maximal Number of Solar Cells in Series	7 pcs
Number of MPPT channels	6 pcs
Maximum Input Power per Channels	12 W
Maximum incoming solar energy	50 W
Energy Storage	
Battery Pack Configuration	2P3S
Nominal Pack Capacity (90% SOC)	5.4 Ah
Nominal Pack Stored Energy (90% SOC)	58.3 Wh
Integrated Battery Heater	YES
Balancer	YES (passive)
Energy Distribution	
Number of LCL channels	12

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Electrical Power System (EPS) / Specification

Available Voltage Levels	3.3 V, 5.0 V, 9.9 V – 12.3 V (unregulated)
Maximum Output Current @3.3V	4 A
Maximum Output Current @5.0V	4 A
Maximum Output Current @UNREG	5 A
Maximum Peak Output Power	90 W
Interfaces	
Connector	Micro-D (MIL-DTL-83513)
Communication	Redundant CAN
Dimensions	
Dimension	88.5 x 96 x 86 mm
Mass	~860g